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RELAXATION OF TURBULENT STRESS
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Equations for the pulsational components of the velocity and temperature yield relaxational formulas for the turbulent stress and the heat flux.

## 1. Relaxation of Turbulent Stress

Attention was first drawn to the analogy between a turbulent flow of Newtonian fluid and a laminar flow of non-Newtonian fluid in [1]. There have been a number of works in which turbulence phenomena are identified with viscoelasticity phenomena [2-4]. In [5], attention turned to the possibility of making use of the pulsational components of the velocity to close the equations for the mean velocities.

From these equations, by well-known means, it is possible to write relaxational formulas for the turbulent stress. To this end, consider the general equations of motion of an incompressible liquid in the absence of external bulk forces

$$
\begin{gather*}
\frac{\rho \partial u_{i}}{\partial t}=-\frac{\partial}{\partial x_{k}}\left(p_{k i}+\rho u_{i} u_{k}\right)  \tag{1}\\
\frac{\partial u_{k}}{\partial x_{k}}=0, \quad i=1,2,3, k=1,2,3 . \tag{2}
\end{gather*}
$$

According to Reynolds' concepts, the hydrodynamic quantities are separated into mean and pulsational components

$$
\begin{equation*}
u_{i}=\bar{u}_{i}+u_{i}^{\prime}, u_{k}=\bar{u}_{k}+u_{k}^{\prime}, p_{k i}=\bar{p}_{k i}+p_{k i}^{\prime} \tag{3}
\end{equation*}
$$

Now the following equations for the mean and pulsational velocities may be derived from Eq. (1) $[6]$

[^0]\[

$$
\begin{gather*}
\frac{\rho \partial \bar{u}_{i}}{\partial t}=-\frac{\partial}{\partial x_{k}}\left(\bar{p}_{k i}+\rho \bar{u}_{i} \bar{u}_{k}+\rho \overline{\rho u_{i}^{\prime}} \overline{u_{k}^{\prime}}\right)  \tag{4}\\
\frac{\rho \partial u_{i}^{\prime}}{\partial t}=-\frac{\partial}{\partial x_{k}}\left[p_{k i}^{\prime}+\rho\left(\overline{u_{i}} u_{k}^{\prime}+u_{i}^{\prime} \bar{u}_{k}\right)+\rho\left(u_{i}^{\prime} u_{k}^{\prime}-\overline{u_{i}^{\prime} u_{k}^{\prime}}\right)\right]  \tag{5}\\
\frac{\rho \partial u_{j}^{\prime}}{\partial t}=-\frac{\partial}{\partial x_{k}}\left[p_{k j}^{\prime}+\rho\left(\bar{u}_{j} u_{k}^{\prime}+u_{j}^{\prime} \bar{u}_{k}\right)+\rho\left(u_{j}^{\prime} u_{k}^{\prime}-\overline{u_{j}^{\prime} u_{k}^{\prime}}\right)\right] . \tag{6}
\end{gather*}
$$
\]

From Eqs. (5) and (6), an expression is obtained for the components of the turbulent-stress tensor $\overline{u_{i}{ }^{\prime} u_{j}}$, bearing in mind the obvious formula

$$
\frac{\partial u_{i}^{\prime} u_{j}^{\prime}}{\partial t}=u_{i}^{\prime} \frac{\partial u_{j}^{\prime}}{\partial t}+u_{j}^{\prime} \frac{\partial u_{i}^{\prime}}{\partial t}
$$

Multiplying Eq. (5) by $u_{j}{ }^{\prime}$ and Eq. (6) by $u_{i}{ }^{\prime}$ and adding them yields the result

$$
\begin{equation*}
\frac{\rho d u_{i}^{\prime} u_{j}^{\prime}}{d t}=-u_{j}^{\prime} \frac{\partial}{\partial x_{k}}\left[p_{k i}^{\prime}+\bar{\rho}_{i} u_{k}^{\prime}+\rho\left(u_{i}^{\prime} u_{k}^{\prime}-\overline{u_{i}^{\prime}} u_{k}^{\prime}\right)\right]-u_{i}^{\prime} \frac{\partial}{\partial x_{k}}\left[p_{k j}^{\prime}+\bar{\rho}_{j} u_{k}^{\prime}+\rho\left(u_{j}^{\prime} u_{k}^{\prime}-\overline{\left.u_{j}^{\prime} u_{k}^{\prime}\right)}\right]\right. \tag{7}
\end{equation*}
$$

The convective-product operator has been introduced here

$$
\frac{d u_{i}^{\prime} u_{j}^{\prime}}{d t}=\frac{\partial u_{i}^{\prime} u_{j}^{\prime}}{\partial t}+\bar{u}_{k} \frac{\partial u_{i}^{\prime} u_{j}^{\prime}}{\partial x_{\hbar}}
$$

A11 the terms in Eq. (7) are now averaged, using the Reynolds equations

$$
\begin{equation*}
\overline{u_{i}^{\prime}}=\overline{u_{k}^{\prime}}=0 ; \overline{\overline{u_{i}^{\prime} u_{k}^{\prime}}}=\overline{u_{i}^{\prime} u_{k}^{\prime}} ; \overline{\overline{u_{k}}}=\overline{u_{k}} \tag{8}
\end{equation*}
$$

and adopting the hypothesis that

$$
\begin{equation*}
\overline{u_{j}^{\prime} \frac{\overline{\partial u_{i}^{\prime} u_{k}^{\prime}}}{\partial x_{k}}}=0 ; u_{i}^{\prime} u_{k}^{\prime} \frac{\partial u_{j}^{\prime}}{\partial x_{k}}=0 \tag{9}
\end{equation*}
$$

Then Eq. (7) takes the form

$$
\begin{align*}
& \left.\frac{\rho \overline{d u_{i}^{\prime} u_{j}^{\prime}}}{d t}=-\left\{\frac{\partial}{\partial x_{k}} \overline{\left\lfloor u_{j}^{\prime}\left(p_{k i}^{\prime}+\rho u_{i}^{\prime} u_{k}^{\prime}\right)\right]}+\frac{\partial}{\partial x_{k}} \overline{\left[u_{i}^{\prime}\left(p_{k j}^{\prime}+\rho u_{j}^{\prime} u_{k}^{\prime}\right)\right.}\right]\right\} \\
& \quad+\overline{p_{k i}^{\prime} \frac{\partial u_{j}^{\prime}}{\partial x_{k}}}+\overline{p_{k i}^{\prime}} \frac{\partial u_{i}^{\prime}}{\partial x_{k}}-\left[\rho \overline{u_{j}^{\prime} u_{k}^{\prime}} \frac{\partial \bar{u}_{i}}{\partial x_{k}}+\overline{\rho u_{i}^{\prime} u_{k}^{\prime}} \frac{\partial \overline{u_{j}}}{\partial x_{k}}\right] . \tag{10}
\end{align*}
$$

Integration over an arbitrary volume $V$ is now performed in Eq. (10), and in the first two terms on the right of the resulting expression the Ostrogradskii-Gauss theorem is used in the following form

$$
\int_{V}\left\{\frac{\partial}{\partial x_{k}} \overline{\left[u_{j}^{\prime}\left(p_{k i}^{\prime}+\rho u_{i}^{\prime} u_{k}^{\prime}\right)\right]}+\frac{\partial}{\partial x_{k}} \overline{\left[u_{i}^{\prime}\left(p_{k j}^{\prime}+\rho u_{j}^{\prime} u_{k}^{\prime}\right)\right.}\right\} d V=\int_{S} n_{k}\left\{\frac{\partial}{\partial x_{k}} \overline{\left[u_{j}^{\prime}\left(p_{k i}^{\prime}+\rho u_{i}^{\prime} u_{k}^{\prime}\right)\right.}+\frac{\partial}{\partial x_{k}} \overline{\left[u_{i}^{\prime}\left(p_{k j}^{\prime}+\rho u_{j}^{\prime} u_{k}^{\prime}\right)\right.}\right\} d S .
$$

Here $n_{k}$ are the directional cosines of the normal to the surface $S$. The surface integral expresses the energy transfer with respect to motion through the surface $S$ and, according to the discussion of Reynolds [6], is equal to zero. Then, the following relation replaces Eq. (10)

$$
\int_{\bar{V}} \overline{\frac{\rho d u_{i}^{\prime} u_{i}^{\prime}}{d t}} d V=\int_{V}\left(\overline{p_{k i}^{\prime} \frac{\partial u_{j}^{\prime}}{\partial x_{k}}}+\overline{p_{k j}^{\prime} \frac{\partial u_{i}^{\prime}}{\partial x_{k}}}\right) d V-\int_{V} \rho\left(\overline{u_{j}^{\prime} u_{k}^{\prime}} \frac{\partial \overline{u_{i}}}{\partial x_{k}}+\overline{u_{i}^{\prime} u_{k}^{\prime}} \frac{\partial \overline{u_{j}}}{\partial x_{k}}\right) d V
$$

But since the elementary volume $V$ is arbitrary, this equation may be rewritten in the form

$$
\begin{equation*}
\frac{\rho \overline{d u_{i}^{\prime} u_{j}^{\prime}}}{d t}=\overline{p_{k i}^{\prime} \frac{\partial u_{j}^{\prime}}{\partial x_{k}}}+\overline{p_{k j}^{\prime} \frac{\partial u_{i}^{\prime}}{\partial x_{k}}}-\rho\left(\overline{u_{j}^{\prime} u_{k}^{\prime}} \frac{\partial \bar{u}_{i}}{\partial x_{k}}+\overline{u_{i}^{\prime} u_{k}^{\prime}} \frac{\partial \overline{u_{i}}}{\partial x_{k}}\right) . \tag{11}
\end{equation*}
$$

Let $i=j$; then Eq. (11) gives

$$
\begin{equation*}
\frac{d \overline{E^{\prime}}}{d t}=\overline{p_{k i}^{\prime} \frac{\partial u_{i}^{\prime}}{\partial x_{k}}}-\rho \overline{u_{i}^{\prime} u_{k}^{\prime}} \frac{\partial \overline{u_{i}}}{\partial x_{k}} \tag{12}
\end{equation*}
$$

where $\bar{E}^{\prime}=1 / 2 \overline{u_{i}{ }^{2}}$ is the kinetic energy of motion along the trajectory of mean motion [6]. Reynolds calls this the "resolving" equation, and has attempted to calculate the critical number Re from it.

For the mean values, by analogy with the usual representation, the following formulas are introduced

$$
\bar{p}_{i j}=\left\{\begin{array}{l}
-\mu_{i j}\left(\frac{\partial \bar{u}_{i}}{\partial x_{j}}+\frac{\partial \bar{u}_{j}}{\partial x_{i}}\right), i \neq j  \tag{13}\\
\bar{p}-2 \mu_{i i}\left(\frac{\partial \bar{u}_{i}}{\partial x_{i}}\right), i=j
\end{array}\right.
$$

Using Eq. (13), the mean-deformation-rate tensor can be expressed in terms of the Maxwellian relaxation time

$$
\begin{align*}
\frac{\partial \bar{u}_{i}}{\partial x_{j}}+\frac{\partial \bar{u}_{j}}{\partial x_{i}} & =-\frac{\bar{p}_{i j}}{\mu_{i j}}=-\frac{1}{\tau_{i j}}, i \neq j \\
2 \frac{\partial \bar{u}_{i}}{\partial x_{i}} & =\frac{\bar{p}_{i j}-\bar{p}}{-\mu_{i i}}=-\frac{1}{\tau_{i i}} \tag{14}
\end{align*}
$$

Introducing the usual formulas for the turbulent stress

$$
\sigma_{i j}=-\rho \overline{\rho u_{i}^{\prime} u_{j}^{\prime}}
$$

Eq. (11) can be rewritten in the form

$$
\begin{equation*}
-\frac{d \sigma_{i j}}{d t}=\sigma_{j k} \frac{\partial \bar{u}_{i}}{\partial x_{k}}+\sigma_{i k} \frac{\partial \bar{u}_{j}}{\partial x_{k}}+\overline{p_{k i}^{\prime} \frac{\partial u_{j}^{\prime}}{\partial x_{k}}}+\overline{p_{h_{j}}^{\prime} \frac{\partial u_{i}^{\prime}}{\partial x_{k}}} \tag{15}
\end{equation*}
$$

The well-known formulas for deformational motion are used

$$
\begin{align*}
& \frac{\partial \bar{u}_{i}}{\partial x_{k}}=\frac{1}{2}\left(\frac{\partial \bar{u}_{i}}{\partial x_{k}}+\frac{\partial \bar{u}_{k}}{\partial x_{i}}\right)+\frac{1}{2}\left(\frac{\partial \bar{u}_{i}}{\partial x_{k}}-\frac{\partial \bar{u}_{k}}{\partial x_{i}}\right),  \tag{16}\\
& \frac{\partial \bar{u}_{j}}{\partial x_{k}}=\frac{1}{2}\left(\frac{\partial \bar{u}_{j}}{\partial x_{k}}+\frac{\partial \bar{u}_{k}}{\partial x_{j}}\right)+\frac{1}{2}\left(\frac{\partial \bar{u}_{j}}{\partial x_{k}}-\frac{\partial \bar{u}_{k}}{\partial x_{j}}\right) .
\end{align*}
$$

Taking these formulas into account, Eq. (15) may be divided into two:

$$
\begin{gather*}
-\frac{d \sigma_{i j}}{d t}=\frac{\sigma_{j_{k}}}{2}\left(\frac{\partial \bar{u}_{i}}{\partial x_{k}}+\frac{\partial \bar{u}_{k}}{\partial x_{i}}\right)+\frac{\sigma_{i k}}{2}\left(\frac{\partial \bar{u}_{j}}{\partial x_{k}}+\frac{\partial \bar{u}_{k}}{\partial x_{j}}\right)  \tag{17}\\
\overline{p_{k i}^{\prime}} \frac{\partial u_{j}^{\prime}}{\partial x_{k}}+p_{k_{j}}^{\prime} \frac{\partial u_{i}^{\prime}}{\partial x_{k}}=-\frac{1}{2}\left(\frac{\partial \bar{u}_{i}}{\partial x_{k}}-\frac{\partial \bar{u}_{k}}{\partial x_{i}}\right)-\frac{1}{2}\left(\frac{\partial \bar{u}_{j}}{\partial x_{k}}-\frac{\partial \bar{u}_{k}}{\partial x_{j}}\right) . \tag{18}
\end{gather*}
$$

Equation (18) expresses the dissipation of relative motion through vorticity of the mean motion.

If the mean velocities are eliminated in Eq. (17), in accordance with Eq. (14), the result is

$$
\begin{gather*}
-\frac{d \sigma_{i j}}{d t}+\frac{1}{2}\left(\frac{1}{\tau_{i i}}+\frac{1}{\tau_{j j}}\right) \sigma_{i j}+\frac{\sigma_{i i}+\sigma_{j j}}{2 \tau_{i j}}+\frac{\sigma_{i k}}{2 \tau_{j k}}+\frac{\sigma_{j_{k}}}{2 \tau_{i k}}=0  \tag{19}\\
i, j=1,2,3 ; k \neq i, k \neq j \\
-\frac{d \sigma_{i i}}{d t}+\sum_{k=1}^{3} \frac{\sigma_{i k}}{\tau_{i k}}=0 \tag{20}
\end{gather*}
$$

The expressions obtained are relaxational formulas for the turbulent stress.
For the purposes of subsequent discussion, Eq. (4) is rewritten in the form

$$
\begin{equation*}
\rho \frac{d \overline{d u}_{i}}{d t}=-\frac{\partial}{\partial x_{\dot{\imath}}}\left(\bar{p}_{k i}-\sigma_{k i}\right) \tag{21}
\end{equation*}
$$

The turbulent stress is eliminated by a means used in deriving heat-conduction equations of hyperbolic type [7]. Both sides of Eq. (21) are differentiated with respect to time and the resulting relation is premultiplied by $\tau_{i i}$, to give

$$
\begin{equation*}
\tau_{i i} \rho \frac{d^{2} \bar{u}_{i}}{d t^{2}}=-\tau_{i i} \frac{\partial}{\partial x_{k}}\left[\frac{d}{d t}\left(\bar{p}_{k i}-\sigma_{k i}\right)\right] . \tag{22}
\end{equation*}
$$

Adding Eqs. (21) and (22) leads to the relation

$$
\begin{equation*}
\tau_{i i} \rho \frac{d^{2} \bar{u}_{i}}{d t^{2}}+\rho \frac{\overline{d u}_{i}}{d t}=-\frac{\partial}{\partial x_{k}}\left[\left(\bar{p}_{k i}-\sigma_{k i}\right)+\tau_{i i} \frac{d}{d t}\left(\bar{p}_{k i}-\sigma_{k i}\right)\right] \tag{23}
\end{equation*}
$$

To close this equation, the following hypothesis is introduced

$$
\left(\bar{p}_{k i}-\sigma_{k i}\right)+\tau_{i i} \frac{d}{d t}\left(\bar{p}_{k i}-\sigma_{k i}\right)=\left\{\begin{array}{c}
-\mu_{k i}\left(\frac{\partial \bar{u}_{k}}{\partial x_{i}}+\frac{\partial \bar{u}_{i}}{\partial x_{k}}\right), k \neq i  \tag{24}\\
\bar{p}-2 \mu_{i i}\left(\frac{\partial \bar{u}_{i}}{\partial x_{i}}\right), k=i
\end{array}\right.
$$

This is analogous to the Maxwellian hypothesis for viscoelastic stress. Now Eq. (23) is rewritten in the form:

$$
\begin{gather*}
\tau_{i i} \rho \frac{d^{2} \bar{u}_{i}}{d t^{2}}+\rho \frac{d \vec{u}_{i}}{d t}=-\frac{\partial \vec{p}}{\partial x_{i}}+\mu_{k i} \frac{\partial^{2} \bar{u}_{i}}{\partial x_{k}^{2}}+\frac{\partial}{\partial x_{i}}\left(\mu_{k i} \frac{\partial \vec{u}_{k}}{\partial x_{k}}\right) \\
i, k=1,2,3 ; \mu_{k i}=\mu_{i k} \tag{25}
\end{gather*}
$$

If the velocities of mean motion are found from Eq. (25), then $\sigma_{i j}$ and $\bar{p}_{k i}$ may be found from Eqs. (19), (20), and (24).

The formula for the second convective product takes the form
$\frac{d^{2}}{d t^{2}}=\frac{\partial^{2}}{\partial t^{2}}+u_{k}^{2} \frac{\partial^{2}}{\partial x_{k}^{2}}+\frac{d \bar{u}_{k}}{d t} \frac{\partial}{\partial x_{k}}+2 \vec{u}_{k} \frac{\partial^{2}}{\partial t \partial x}+2 \bar{u}_{1} \bar{u}_{2} \frac{\partial^{2}}{\partial x_{1} \partial x_{2}}+2 \bar{u}_{1} \bar{u}_{3} \frac{\partial^{2}}{\partial x_{1} \partial x_{3}}+2 \bar{u}_{2} \bar{u}_{3} \frac{\partial^{2}}{\partial x_{2} \partial x_{3}} \cdot(26$
In the simplest case, when $\tau_{i j}=\tau$, $\mu_{k i}=\mu$, Eq. (25) takes the simpler form

$$
\begin{equation*}
\tau \frac{d^{2} \bar{u}_{i}}{d t^{2}}+\frac{d \bar{u}_{i}}{d t}=-\frac{1}{\rho} \frac{\partial \bar{p}}{\partial x_{i}}+v \nabla^{2} \bar{u}_{i} \tag{27}
\end{equation*}
$$

## 2. Relaxation of Heat Flux

Consider the heat-balance equation in a turbulent flow of incompressible liquid, disregarding energy dissipation

$$
\begin{equation*}
c_{0} \frac{d T}{d t}=-\frac{\partial q_{k}}{\partial x_{k}} \tag{28}
\end{equation*}
$$

Here $c_{0}$ is the product of the specific heat and the density. Separating the quantities appearing in Eq. (28) into mean and pulsational components

$$
\begin{equation*}
T=\bar{T}+T^{\prime}, \quad u_{k}=\bar{u}_{k}+u_{k}^{\prime}, q_{k}=\bar{q}_{k}+q_{k}^{\prime} \tag{29}
\end{equation*}
$$

and performing the usual averaging operation for all the terms in Eq. (28) leads to the result

$$
\begin{equation*}
c_{0} \frac{d \bar{T}}{d t}=-\frac{\partial}{\partial x_{k}}\left(\bar{q}_{k}+c_{0} \overline{u_{k}^{\prime} T^{\prime}}\right) \tag{30}
\end{equation*}
$$

Subtracting Eq. (30) from Eq. (28) gives

$$
\begin{equation*}
c_{0} \frac{d T^{\prime}}{d t}=-\frac{\partial}{\partial x_{k}}\left[q_{k}^{\prime}+c_{0} u_{k}^{\prime} \bar{T}+c_{0}\left(u_{k}^{\prime} T^{\prime}-\overline{u_{k}^{\prime} T^{\prime}}\right)\right] \tag{31}
\end{equation*}
$$

By the above-outlined means, the following equation may be obtained from Eqs. (31) and (5):

$$
\begin{equation*}
\overline{\frac{d u^{\prime} T^{\prime}}{d t}}=\frac{1}{\rho} \overline{p_{k i}^{\prime} \frac{\partial T^{\prime}}{\partial x_{k}}}+\frac{1}{c_{0}} \overline{q_{k}^{\prime} \frac{\partial u_{i}^{\prime}}{\partial x_{k}}}-\left(\overline{T^{\prime} u_{k}^{\prime}} \frac{\partial \overline{u_{i}}}{\partial x_{k}}+\overline{u_{i}^{\prime} u_{k}^{\prime}} \frac{\partial \bar{T}}{\partial x_{k}}\right) . \tag{32}
\end{equation*}
$$

It has been assumed that

$$
\overline{u_{i}^{\prime} \frac{\overline{\partial u_{k}^{\prime} T^{\prime}}}{\partial x_{k}}}=0, \overline{u_{k}^{\prime} T^{\prime} \frac{\partial u_{i}^{\prime}}{\partial x_{k}}}=0
$$

If Eq. (16) is used, Eq. (32) may be replaced by two relations:

$$
\begin{gather*}
-\frac{d \overline{u_{i}^{\prime} T^{\prime}}}{d t}+\frac{\overline{u_{k}^{\prime} T^{\prime}}}{2}\left(\frac{\partial \bar{u}_{i}}{\partial x_{k}}+\frac{\partial \overline{u_{k}}}{\partial x_{i}}\right)=0  \tag{33}\\
\frac{1}{c_{0}} \overline{q_{k}^{\prime} \frac{\partial u_{i}^{\prime}}{\partial x_{k}}}+\frac{1}{\rho} \overline{p_{k i}^{\prime} \frac{\partial T^{\prime}}{\partial x_{k}}}=\overline{u_{i}^{\prime} u_{k}^{\prime}} \frac{\partial \bar{T}}{\partial x_{k}}+\overline{\frac{u_{k}^{\prime} T^{\prime}}{2}}\left(\frac{\partial \bar{u}_{i}}{\partial x_{k}}-\frac{\partial \bar{u}_{k}}{\partial x_{i}}\right) . \tag{34}
\end{gather*}
$$

In (33), the mean values may be expressed in terms of the relaxation-time spectrum in Eq. (14), when

$$
\begin{equation*}
\frac{d \overline{u_{i}^{\prime} \overline{T^{\prime}}}}{d t}-\frac{\overline{u_{h}^{\prime} T^{\prime}}}{2 \tau_{i h}}=0 \tag{35}
\end{equation*}
$$

After differentiation with respect to the time, Eq. (30) is rewritten in the form

$$
\tau_{i i} c_{0} \frac{d^{2} \bar{T}}{d t^{2}}=-\tau_{i i} \frac{\partial}{\partial x_{k}}\left[\frac{d}{d t}\left(\bar{q}_{k}+c_{0} \overline{u_{k}^{\prime} T^{\prime}}\right)\right] .
$$

The resulting relation is combined with Eq. (30)

$$
\tau_{i i} c_{0} \frac{d^{2} \bar{T}}{d t^{2}}+c_{v} \frac{d \bar{T}}{d t}=-\frac{\partial}{\partial x_{k}}\left[\left(\bar{q}_{k}+c_{0} \overline{u_{k}^{\prime} T^{\prime}}\right)+\tau_{i i}-\frac{d}{d t}\left(\bar{q}_{k}+c_{0} \overline{u_{k}^{\prime} T^{\prime}}\right)\right]
$$

and the following hypothesis is adopted

$$
\begin{equation*}
\left.\bar{q}_{k}+c_{0} \overline{u_{k}^{\prime} T^{\prime}}\right)+\tau_{i i} \frac{d}{d t}\left(\bar{q}_{h}+c_{0} \overline{u_{h}^{\prime} T^{\prime}}\right)=-\lambda_{k} \frac{\partial \bar{T}}{\partial x_{k}} . \tag{36}
\end{equation*}
$$

The result obtained is

$$
\begin{equation*}
\tau_{i i} \frac{d^{2} \bar{T}}{d t^{2}}+\frac{d \bar{T}}{d t}=k_{i} \frac{\partial^{2} \bar{T}}{\partial x_{i}^{2}} . \tag{37}
\end{equation*}
$$

If Eq. (37) is solved, then $\overline{\mathrm{q}_{\mathrm{k}}}$ and $\overline{\mathrm{u}_{\mathrm{i}}{ }^{\prime} \mathrm{T}}$ may be determined from Eqs. (35) and (36).

## 3. Problem of Accelerating Flow Close to a Plane Wall

Suppose that a plane wall, previously at rest, suddenly_begins_to move in its own plane at a constant velocity $U_{0}$. For such a flow, $\bar{u}_{1}=\bar{u}_{1}\left(x_{2}, t\right), \bar{u}_{2}=0, \bar{u}_{3}=0$, and the pressure remains constant [8]. With these assumptions, Eq. (27) is simplified:

$$
\begin{equation*}
\tau \frac{\partial^{2} U}{\partial t^{2}}+\frac{\partial U}{\partial t}=v \frac{\partial^{2} U}{\partial y^{2}} . \tag{38}
\end{equation*}
$$

Here $U(t, y)=\bar{u}_{1}\left(x_{2}, t\right)$. The initial conditions are

$$
\begin{gather*}
\text { when } t \leqslant 0 \quad U=0 \text { for all } y,  \tag{39}\\
\text { when } t>0 \quad U=U_{0} \quad \text { for } y=0, U=0, y=\infty .
\end{gather*}
$$

Let $\sigma=\bar{p}_{12}-\sigma_{12}$ denote the difference between the mean and turbulent stresses. Then Eq. (24) may be rewritten for this problem in the form:

$$
\begin{equation*}
\sigma+\tau \frac{\partial \sigma}{\partial t}=-\mu \frac{\partial U}{\partial y} . \tag{40}
\end{equation*}
$$

It is now expedient to convert to dimensionless values by means of the formulas

$$
t=\tau \bar{t}, y=\sqrt{\bar{v}} \bar{y}, U=U_{0} \bar{U}, \sigma=\sigma_{m} \bar{\sigma} .
$$

It may readily be established that Eqs. (38) and (39) are equivalent to the system of equations

$$
\begin{equation*}
\frac{\partial U}{\partial y}=-\left(\frac{\partial \sigma}{\partial t}+\sigma\right), \frac{\partial U}{\partial t}=-\frac{\partial \sigma}{\partial y} \tag{41}
\end{equation*}
$$

in which, for convenience, the bars above the dimensionless quantities have been omitted.
This system of equations was studied by Gon'o in connection with the theory of the hyperbolic heat-conduction equation, and its results are outlined in detail in [7]. According to this, it is found that

TABLE 1. Numerical Values of Frictional Stress at Wall


Consider the value of the stress $\sigma$ at the wall:

$$
\begin{equation*}
\sigma(t, 0)=\sigma_{0}(t)=e^{-\frac{t}{2}} I_{0}\left(\frac{1}{2} t\right) \tag{43}
\end{equation*}
$$

At sufficiently large $t$, using the asymptotic expansion for the Bessel function $I_{0}$, it may be shown that $\sigma_{0}(t)=1 / \sqrt{\pi t}$. The theory of the Navier-Stokes equation leads exactly to this result [8], which is paradoxical when $t \rightarrow 0$, since $\sigma_{0}(t) \rightarrow \infty$. From Eq. (43), $\sigma_{0}(0)=1$. Hence, the dimensional stress $\sigma$ at the initial instant is

$$
\sigma=\sigma_{m}=\frac{\rho U_{0} \sqrt{v}}{\sqrt{\tau}}
$$

It is simple to see that this "resting stress" must be overcome by the wall if the given mode of flow is to arise.

The idea here outlined leads to the assertion that the flow is initially turbulent but, at some point in time, it becomes laminar if the initial and boundary conditions are retained in the form in Eq. (39). The moment of transition is determined by the relaxational properties of the medium.

Table 1 gives numerical values of the frictional stress at the wall $\sigma_{0}(t)$. It follows from an analysis of these data that discrepancy between Eq. (43) and the results of the theory of the Navier-Stokes equations is observed at a sufficiently large interval of dimensionless time $t$.

## NOTATION

$u_{i}$, component of hydrodynamic velocity; $\bar{u}_{i}$, mean component of velocity; $u_{i}$ ', pulsational component of velocity; pki, stress tensor; pki, mean stress tensor; pki', pulsational stress tensor; $\rho$, density; $T$, temperature; $q_{k}$, heat-flux component; $I_{o}$, Bessel function of zeroth argument.

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